
Titanium & its Alloys

Pure Titanium

Pure titanium melts at 1670°C and has a density of 4.51 g cm^{-3} . It should therefore be ideal for use in components which operate at elevated temperatures, especially where large strength to weight ratios are required. Titanium can catch fire and cause severe damage in circumstances where it rubs against other metals at elevated temperatures (Fig. 1). This is what limits its application in the harsh environment of aeroengines, to regions where the temperature does not exceed 400°C .

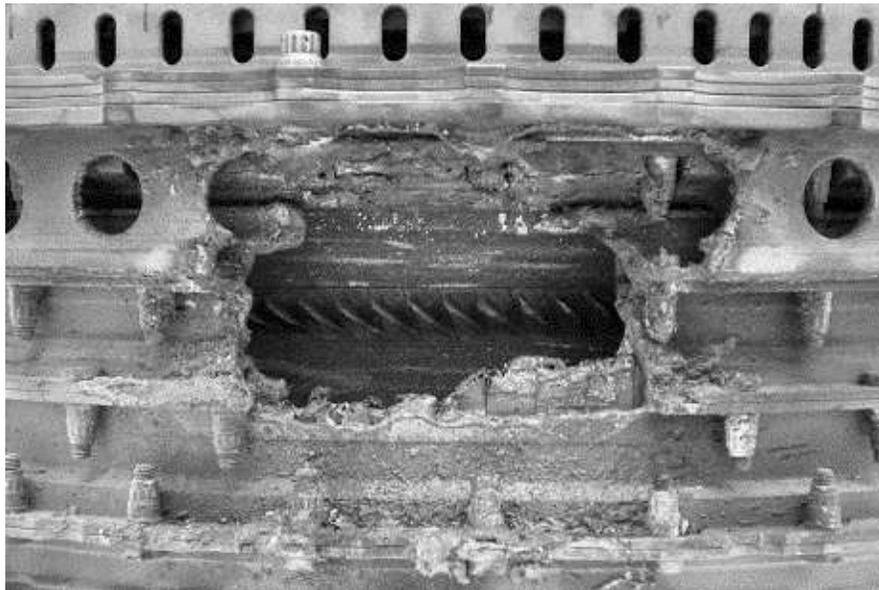


Fig. 1: Consequences of a titanium fire in an aero-engine. Nickel alloy blades have been burnt away. Photograph courtesy of Dr M. Hicks, RR.

The world production of titanium is nevertheless very small, hundreds of thousands of tonnes, which compares say with steel at 750 million tonnes per annum. 80% of all the titanium produced is used in the aerospace industries. Car suspension springs could easily be made of titanium with a great reduction in weight but titanium is not available in the large quantities needed and certainly not at the price required for automobile applications. The target price for titanium needs to be reduced to about 30% of its current value for serious application in mass-market cars.

Pure titanium has excellent resistance to corrosion and is used widely in the chemical industries. There is a passive oxide film which makes it particularly resistant to corrosion in oxidising solutions. The corrosion resistance can be further improved by adding palladium (0.15 wt%), which makes hydrogen evolution easier at cathodic sites so that the anodic and cathodic reactions balance in the passive region (Fig. 2).

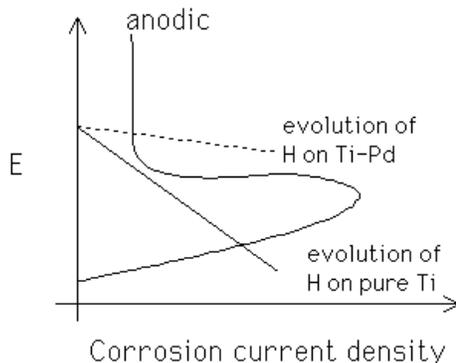


Fig. 2: Corrosion of Ti and Ti-Pd

Most chemical plant use steel vessels which are clad with titanium. The titanium is frequently explosion bonded. Titanium condenser tubes

are used in power plant and in desalination plant.

The crystal structure of titanium at ambient temperature and pressure is close-packed hexagonal (α) with a $\frac{c}{a}$ ratio of 1.587. Slip is possible on the pyramidal, prismatic and basal planes in the close-packed directions (Fig. 3). At about 890 °C, the titanium undergoes an allotropic transformation to a body-centred cubic β phase which remains stable to the melting temperature.

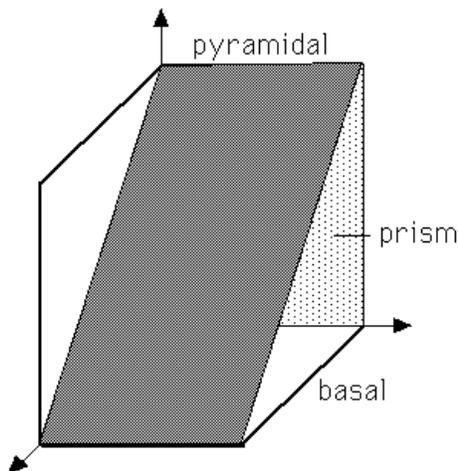


Fig. 3: The slip planes in titanium

Alloying of Ti

All elements which are within the range 0.85–1.15 of the atomic radius of titanium alloy substitutionally and have a significant solubility in titanium. Elements with an atomic radius less than 0.59 that of Ti occupy interstitial sites and also have substantial solubility (*e.g.* H, N, O, C). The ease with which solutes dissolve in titanium makes it difficult to design precipitation-hardened alloys. Boron has a similar but larger radius than C, O, N and H; it is therefore possible to induce titanium

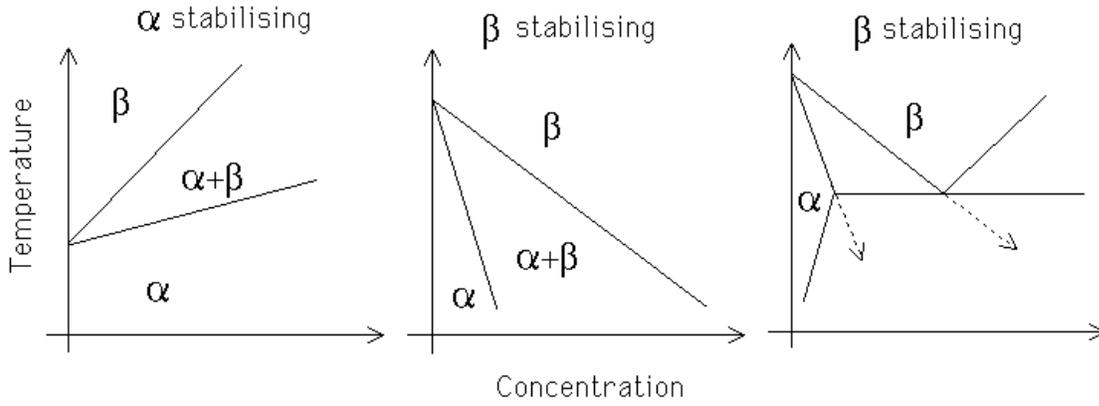
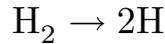


Fig. 4: Phase diagrams for Ti alloys

Interstitials

These do not fit properly and cause changes in the lattice parameters. Hydrogen is the most important interstitial. Body-centred cubic Ti has three octahedral interstices per atom whereas c.p.h. Ti has one per atom. The latter are therefore larger, so that the solubility of O, N, and C is much higher in the α phase.

Titanium is capable of absorbing up to 60 at% of hydrogen, which can also be removed by annealing in a vacuum. Hydrogen enters the tetrahedral holes which are larger in b.c.c. than c.p.h. Thus the solubility of hydrogen is larger in β . The enthalpy of solution of hydrogen in Ti is negative ($\Delta H < 0$).



$$\Delta G = \Delta G_0 + RT \ln \frac{a_{\text{H}}^2}{p_{\text{H}_2}}$$

so that

$$\frac{a_{\text{H}}^2}{p_{\text{H}_2}} = \exp\left\{\frac{\Delta S}{R}\right\} \exp\left\{-\frac{\Delta H}{RT}\right\}$$

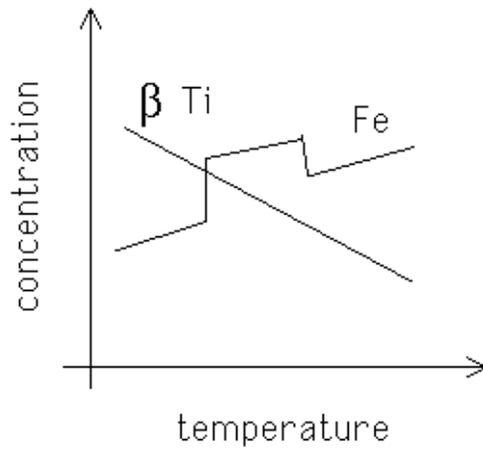


Fig. 5: Solubility of hydrogen

The solubility actually decreases with temperature (Fig. 5). This contrasts with iron which shows the opposite trend.

Because of this characteristic, titanium is a candidate material for the first wall of magnetically confined fusion reactors. The hydrogen based plasma is not detrimental since at 500 °C and 1 Pa pressure, the Ti does not pick up enough hydrogen for embrittlement. An additional feature is that Ti resists swelling due to neutron damage.

A large enough concentration of hydrogen induces the precipitation of hydrides. $\text{TiH}_{1.5-2.0}$ has a Cubic-F lattice and its precipitation causes embrittlement due to a volume expansion of about 18%. There are regions of hydrostatic tension at crack tips where it forms preferentially, leading to large increases in the crack growth rate, some 50-fold during fatigue.

The hydride reaction can also be used to store hydrogen reversibly;



The energy to weight ratio for such a cell is about a tenth that of petrol.

One problem with this method of hydrogen storage is that hydride formation is accompanied by a considerable volume expansion, which in turn can embrittle the alloy. Amorphous alloys of titanium are better in this respect, since they do form hydrides and yet reversibly accommodate large quantities of hydrogen by an expansion of the nearest-neighbour distance.

Zr-Ti Laves phase $\text{Ti}_{0.24}\text{Zr}_{0.76}(\text{Ni}_{0.55}\text{Mn}_{0.3}\text{V}_{0.065}\text{Fe}_{0.085})_{2.1}$ has been found to reversibly accommodate nearly 1.5 wt% of hydrogen, with a battery rating of some 440 mAh g^{-1} .

Specific-alloys

α -alloys

Aluminium is the main alloying element apart from Zr and Sn. The combined effect is expressed as

$$\text{aluminium equivalent, wt\%} = \text{Al} + \frac{1}{3}\text{Sn} + \frac{1}{6}\text{Zr} + 10(\text{O} + \text{C} + 2\text{N})$$

If this exceeds about 9 wt% then there may be detrimental precipitation reactions (generally Ti_3X which has an ordered h.c.p. structure).

The presence of a small amount of the more ductile β -phase in nearly α alloys is advantageous for heat treatment and the ability to forge. The alloys may therefore contain some 1 wt% of Mo *e.g.*



where the Zr and Sn give solid solution strengthening.

A near- α alloy has been developed, with good elevated temperature properties ($T < 590^\circ\text{C}$):



The niobium is added for oxidation resistance and the carbon to allow a greater temperature range over which the alloy is a mixture of $\alpha + \beta$, in order to facilitate thermomechanical processing. This particular alloy is used in the manufacture of aeroengine discs and has replaced discs made from much heavier nickel base superalloys. The final microstructure of the alloy consists of equiaxed primary- α grains, Widmanstätten α plates separated by the β -phase.

$\alpha + \beta$ alloys

Most $\alpha + \beta$ alloys have high-strength and formability, and contain 4–6 wt% of β -stabilisers which allow substantial amounts of β to be retained on quenching from the $\beta \rightarrow \beta + \alpha$ phase fields, *e.g.*



Al reduces density, stabilises and strengthens α while vanadium provides a greater amount of the more ductile β phase for hot-working. This alloy, which accounts for about half of all the titanium that is produced, is popular because of its strength (1100 MPa), creep resistance at 300 °C, fatigue resistance and castability.

One difficulty with the β phase, which has a body-centred cubic crystal structure, is that like ferritic iron, it has a ductile–brittle transition temperature. The transition temperature tends to be above room temperature, with cleavage fracture dominating at ambient temperatures.

Burn-resistant β -alloys

Titanium fires can occasionally occur in aeroengines or in titanium-based heat exchangers used in the chemical industries.

The addition of chromium in concentrations exceeding 10 wt% helps improve the burn-resistance of titanium alloys. The alloy Ti-35V-15Cr wt%, has sufficient chromium to resist burning in an aeroengine environment to temperatures upto about 510 °C. The chromium is not found to be effective in binary Ti-Cr alloys where it does not encourage the formation of a continuous film of protective oxide.

Quenching from β

Quenching the β phase leads to the formation of h.c.p. α' martensite. This is not particularly hard and there are increasing quantities of retained- β in the microstructure as the solute concentration increases and the M_S temperature decreases (Fig. 6).

$$(1\ 1\ 0)_\beta \parallel (0\ 0\ 0\ 1)_{\alpha'} \quad [1\ \bar{1}\ 1]_\beta \parallel [1\ 1\ \bar{2}\ 0]_{\alpha'}$$

and the habit plane of the martensite is close to $\{3\ 3\ 4\}_\beta$.

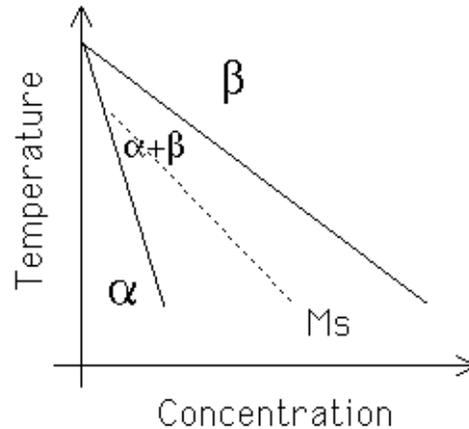


Fig. 6: Martensitic transformation from β

$\beta \rightarrow \omega$ Transformation

ω is a metastable phase which forms from β in alloys based on titanium, zirconium and hafnium. It is important because its formation generally leads to a deterioration in the mechanical properties. In Ti–Nb alloys its formation influences superconduction. The transformation to ω is diffusionless, occurs below the T_0 temperature and frequently cannot be suppressed even by quenching at 11000 C s^{-1} . Its presence causes diffuse streaking in the electron diffraction patterns of the β phase. The streaks become more intense and curved as the temperature or the solute concentration increases. There is also an increase in the electrical resistance as ω forms.

The $\beta \rightleftharpoons \omega$ transformation is reversible and diffusionless but is not martensitic in the classical sense since there is no invariant–plane strain shape deformation. However, it does involve the coordinated motion of atoms.

The body–centred cubic (bcc) crystal structure of β can be imagined as the stacking of $\{111\}_\beta$ planes in an $\dots ABCABC\dots$ stacking sequence. Note that these planes are not close–packed in the bcc structure. The $\beta \rightleftharpoons \omega$ transformation occurs by the passage of a longitudinal displacement wave along $\langle 111 \rangle$ which causes the B and C planes to collapse into each other, leaving the A planes unaffected. The stacking sequence thus changes to $\dots AB'AB'AB'\dots$ in which the B' planes have twice the density of atoms as the A planes. The $\dots AB'AB'AB'\dots$ stacking is consistent with a ω a hexagonal crystal structure with a $c/a \simeq 0.6$. The atoms in the B' plane have a trigonal coordination which is similar to that in graphite and the bonding becomes partly covalent. This leads

to an increase in the electrical resistivity. The longitudinal displacement waves are responsible for the streaking in the electron diffraction patterns.

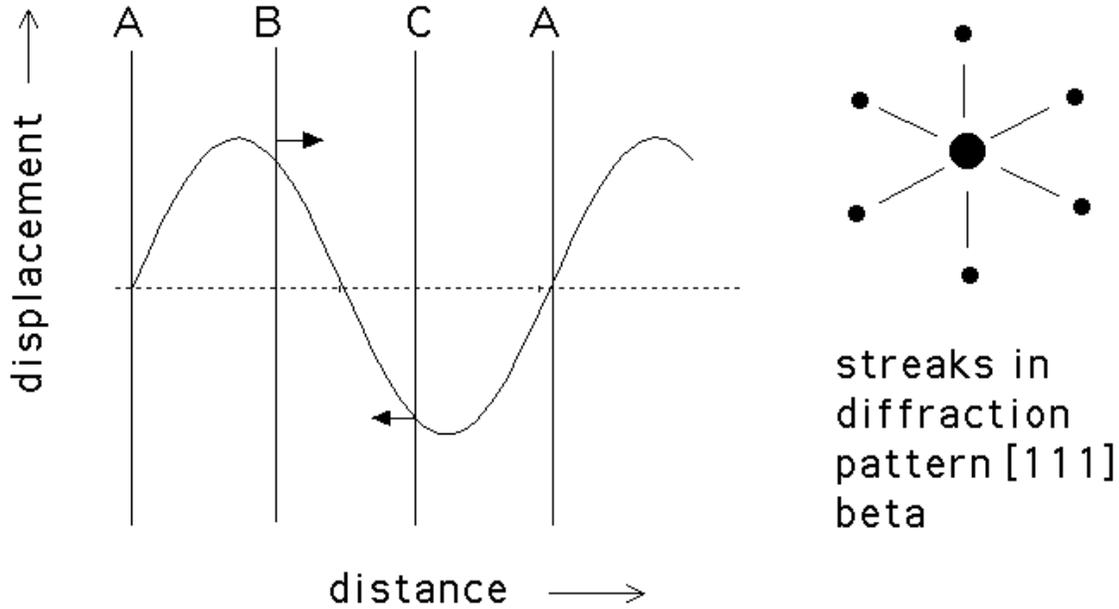


Fig. 6: (a) Displacement wave associated with the β to ω transformation. The A planes are unaffected since they lie at the nodes. (b) Streaks in the electron diffraction pattern during the ω transformation.

Titanium Aluminides

The most successful of the aluminides has a lamellar structure made up of alternating layers of an ordered hexagonal (D_{19}) Ti_3Al and α_2 compound and ordered f.c.t. ($L1_0$) $TiAl$ or γ (Fig. 7). Ductility about 4-6% at ambient temperature. The γ aluminide tends to be more ductile. Density about 4.5 g cm^3 and the aluminium makes them resistant to

burning. The alloys have been extensively studied for aerospace and automotive turbochargers because of their high strength, low density and creep resistance. The γ phase forms with its most closely packed plane parallel to the basal plane of the α_2 :

$$\begin{aligned} \{1\ 1\ 1\}_{\gamma} &\parallel \{0\ 0\ 0\ 1\}_{\alpha_2} \\ \langle 1\ 1\ 0 \rangle_{\gamma} &\parallel \langle 1\ 1\ \bar{2}\ 0 \rangle_{\alpha_2} \end{aligned}$$

The lamellar microstructure is a direct consequence of this orientation relationship.

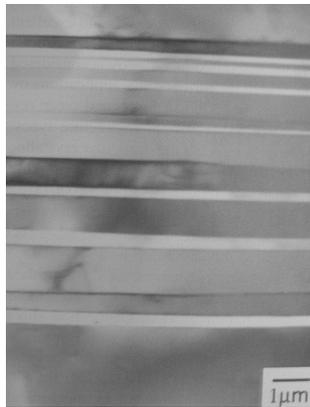


Fig. 7: Ti-48Al: lamellar microstructure (Kim and Maruyama, 2001)

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